

**SIEMENS**



**ARRAYCON - ESR1**

**Smart structures modelling and model order reduction techniques for feasible distributed active damping solutions**

**ZhongZhe DONG**

# Future activities and difficulties

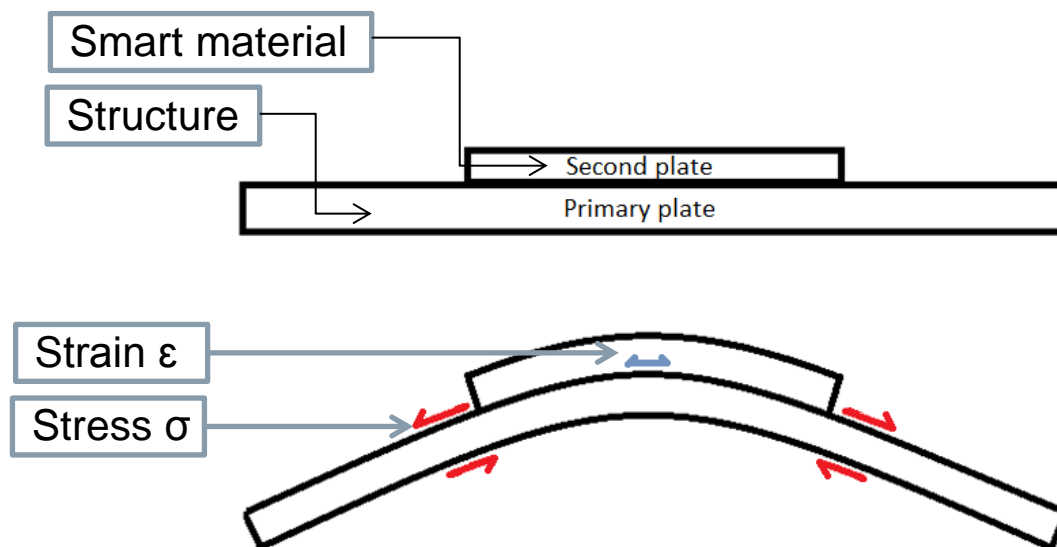
## PAPER SUMMARY:

1. Introduction  
Citing references and the description for the use of this method
2. Neutral surface location for a wafer bonded to a plate  
The arguments of the method to find the new neutral surface in the section of the structure, relation with the existed method
3. Smart materials' strain variation analysis  
Focus on the bending moment, to show how to estimate the curvature and the strain
4. Results comparison and discussion  
some FEM simulation results compare with the analytical results  
comparison between the analytical results with the references results
5. Conclusion

## Recent work

### DONE:

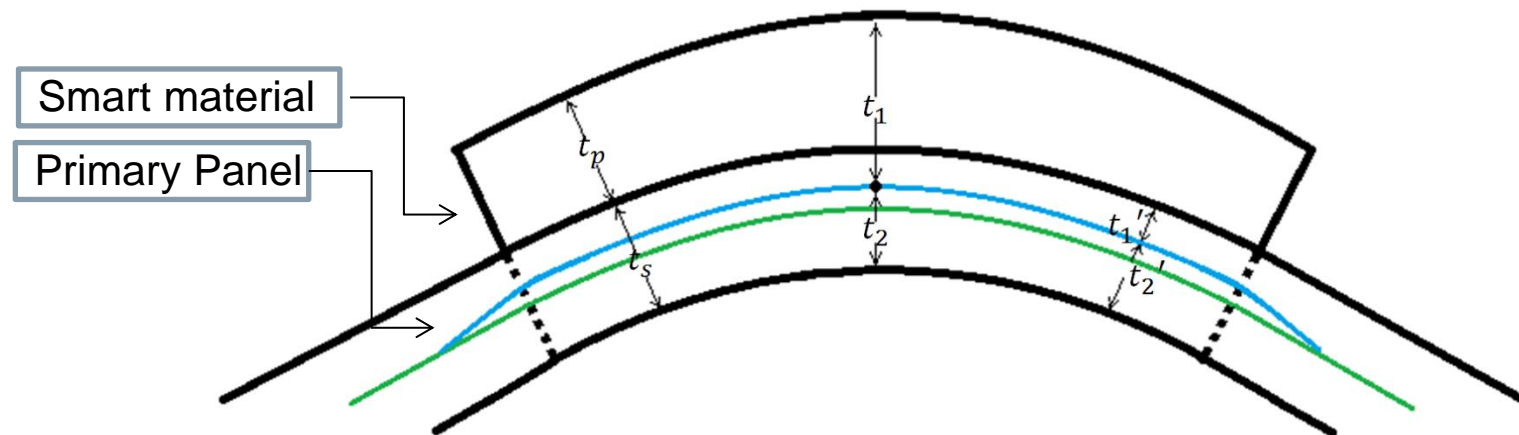
1. Mechanical coupling analysis between primary panel and smart material
2. Mechanical coupling simulations



## Recent work

### MECHANICAL ANALYSIS:

Kirchhoff's plate hypothesis satisfied!!!



- Existence of the neutral axis in the primary panel
- Retention of the neutral axis in the primary panel section
- Smart material in tension and bending
- Exaltation of the neutral axis' location

## Recent work

### MECHANICAL ANALYSIS:

Determine the neutral axis location

**The normal efforts on the neutral axis = null**

$$N_x = -\frac{E^+ t_1'^2}{2(1 - (\vartheta^+)^2)} \left( \frac{\partial^2 w}{\partial x^2} + \vartheta^+ \frac{\partial^2 w}{\partial y^2} \right) - \frac{E_p [(t_1' + t_p)^2 - t_1'^2]}{2(1 - (\vartheta_p)^2)} \left( \frac{\partial^2 w}{\partial x^2} + \nu_p \frac{\partial^2 w}{\partial y^2} \right) + \frac{E^- t_2^2}{2(1 - (\vartheta^-)^2)} \left( \frac{\partial^2 w}{\partial x^2} + \vartheta^- \frac{\partial^2 w}{\partial y^2} \right) = 0$$

$$N_y = -\frac{E^+ t_1'^2}{2(1 - (\vartheta^+)^2)} \left( \frac{\partial^2 w}{\partial y^2} + \vartheta^+ \frac{\partial^2 w}{\partial x^2} \right) - \frac{E_p [(t_1' + t_p)^2 - t_1'^2]}{2(1 - (\vartheta_p)^2)} \left( \frac{\partial^2 w}{\partial y^2} + \nu_p \frac{\partial^2 w}{\partial x^2} \right) + \frac{E^- t_2^2}{2(1 - (\vartheta^-)^2)} \left( \frac{\partial^2 w}{\partial y^2} + \vartheta^- \frac{\partial^2 w}{\partial x^2} \right) = 0$$

After the simplification

$$\frac{E_s (t_2^2 - t_1'^2)}{2(1 - \nu_s)} = \frac{E_p [(t_1' + t_p)^2 - t_p^2]}{2(1 - \nu_p)}$$

$$\text{with } t = t_p + t_s, = t_1 + t_2 \text{ \& } t_s = t_1' + t_2$$

Finally

$$t_1' = \frac{\alpha t_s^2 - \beta t_p^2}{2(\beta t_p + \alpha t_s)}, \text{ which, } \alpha = \frac{1 - \nu_p}{1 - \nu_s} \text{ \& } \beta = \frac{E_p}{E_s}$$

So

$$t_1 = t_p + t_1' \text{ \& } t_2 = t - t_1$$

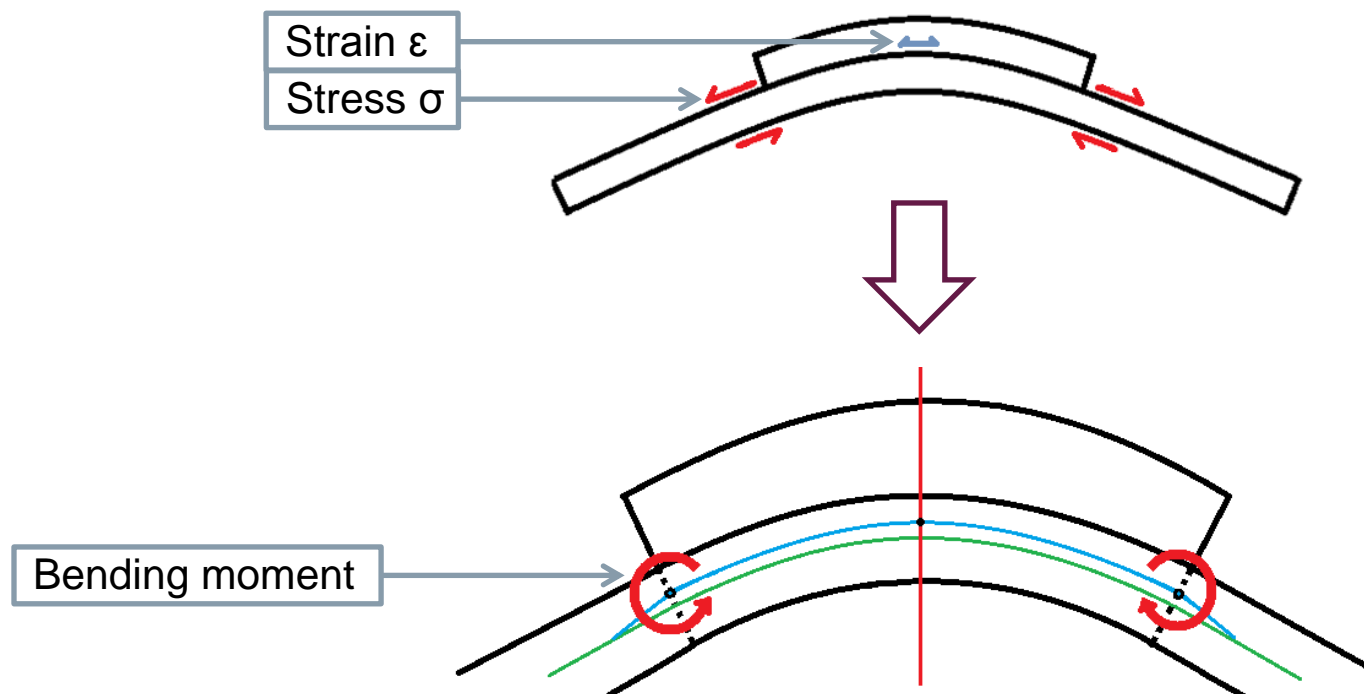
Test case:  $\alpha = \beta = 1$  (Isotropic homogenous materials):  $t_1 = t_2 = \frac{t}{2}$

## Recent work

### MECHANICAL ANALYSIS:

Determine the strain in the smart material

**Conservation of the moments in x and y directions**



## Recent work

### MECHANICAL ANALYSIS:

$$M_x: -a \frac{E^+ t_1^3}{3(1-(\vartheta^+)^2)} (\theta_x + \vartheta^+ \theta_y) - a \frac{E^- t_2^3}{3(1-(\vartheta^-)^2)} (\theta_x + \vartheta^- \theta_y) - a' \frac{E_p [(t_1' + t_p)^3 - t_1'^3]}{3(1-(v_p)^2)} (\theta_x + v_p \theta_y) = -a \frac{E_s t^3}{12(1-\nu^2)} (\theta_x^s + \vartheta \theta_y^s)$$

$$M_y: -b \frac{E^+ t_1^3}{3(1-(\vartheta^+)^2)} (\theta_y + \vartheta^+ \theta_x) - b \frac{E^- t_2^3}{3(1-(\vartheta^-)^2)} (\theta_y + \vartheta^- \theta_x) - b' \frac{E_p [(t_1' + t_p)^3 - t_1'^3]}{3(1-(v_p)^2)} (\theta_y + v_p \theta_x) = -b \frac{E_s t^3}{12(1-\nu^2)} (\theta_y^s + \vartheta \theta_x^s)$$

$$\theta_x = \frac{\partial^2 w}{\partial x^2} \quad \& \quad \theta_y = \frac{\partial^2 w}{\partial y^2} : \text{curvature of the neutral layer in x \& y directions}$$

With

$$\varepsilon_{xx} = -z \frac{\partial^2 w}{\partial x^2} = -z \theta_x$$

$$\varepsilon_{yy} = -z \frac{\partial^2 w}{\partial y^2} = -z \theta_y$$

Finally,

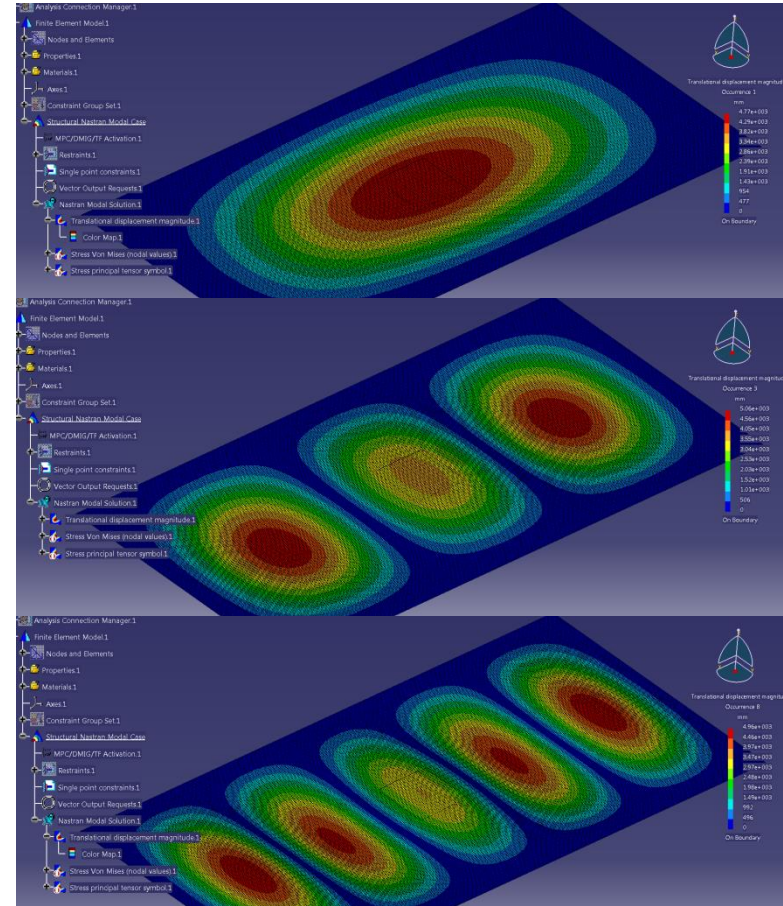
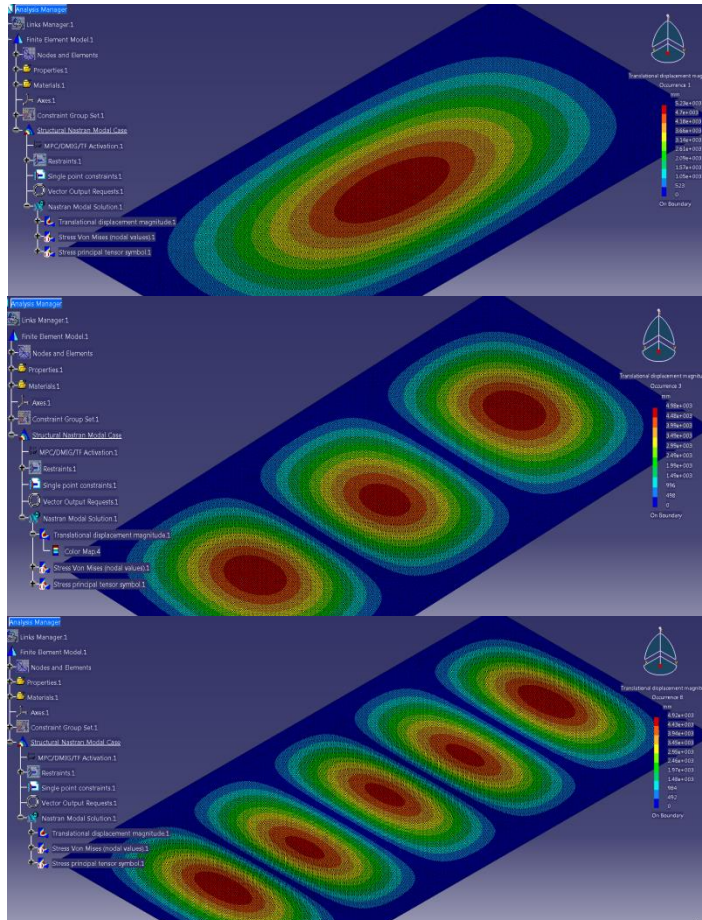
$$\varepsilon_{xx} = \frac{2z}{t_s} \left[ \frac{(aD\gamma_1 - bD\gamma_3\nu_s)\varepsilon_{xx}^s + (aD\gamma_2\nu_s - bD\gamma_3)\varepsilon_{yy}^s}{(\gamma_1\gamma_2 - \gamma_3\gamma_4)} \right]$$

$$\varepsilon_{yy} = \frac{2z}{t_s} \left[ \frac{(aD\gamma_2 - bD\gamma_1\nu_s)\varepsilon_{xx}^s + (aD\gamma_4\nu_s - bD\gamma_1)\varepsilon_{yy}^s}{(\gamma_3\gamma_4 - \gamma_1\gamma_2)} \right]$$

$\gamma_1, \gamma_2, \gamma_3, \gamma_4$ : factors from the Moments equations

# Recent work

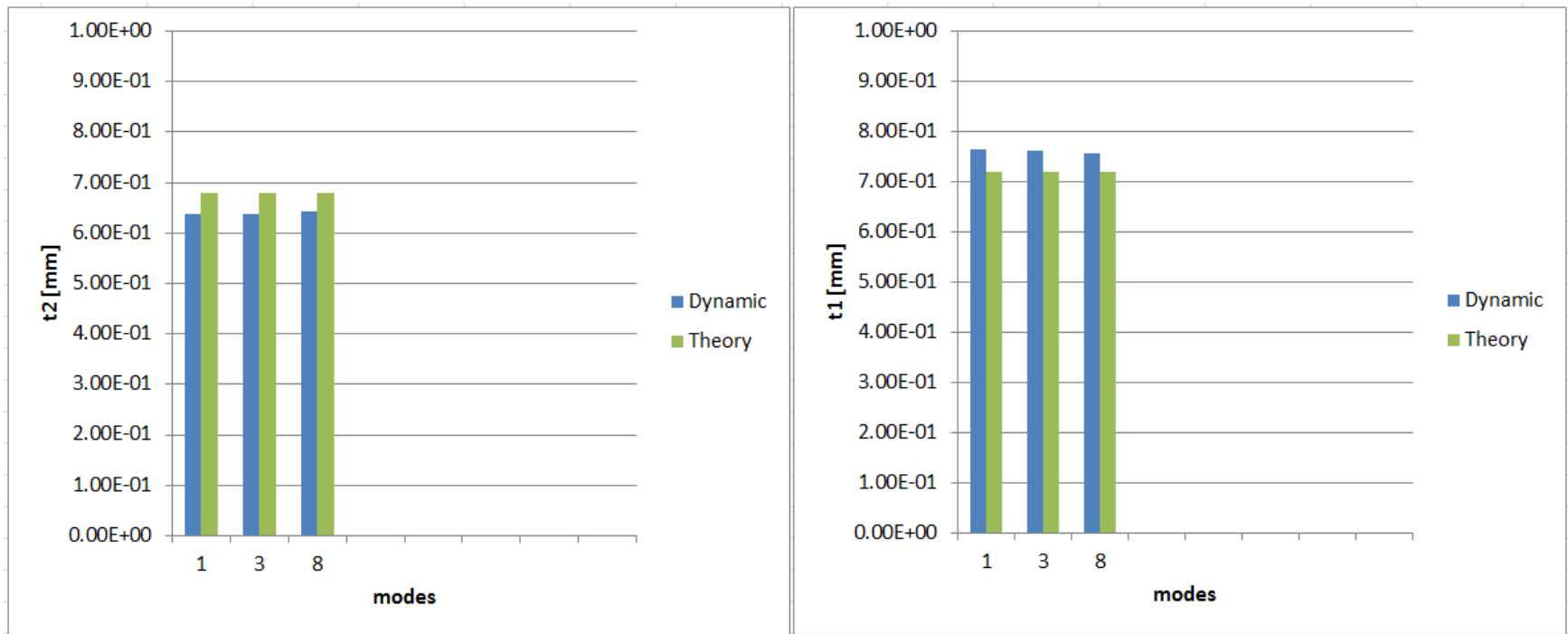
## ONE SMART MATERIAL NUMERICAL SIMULATIONS:





# Recent work

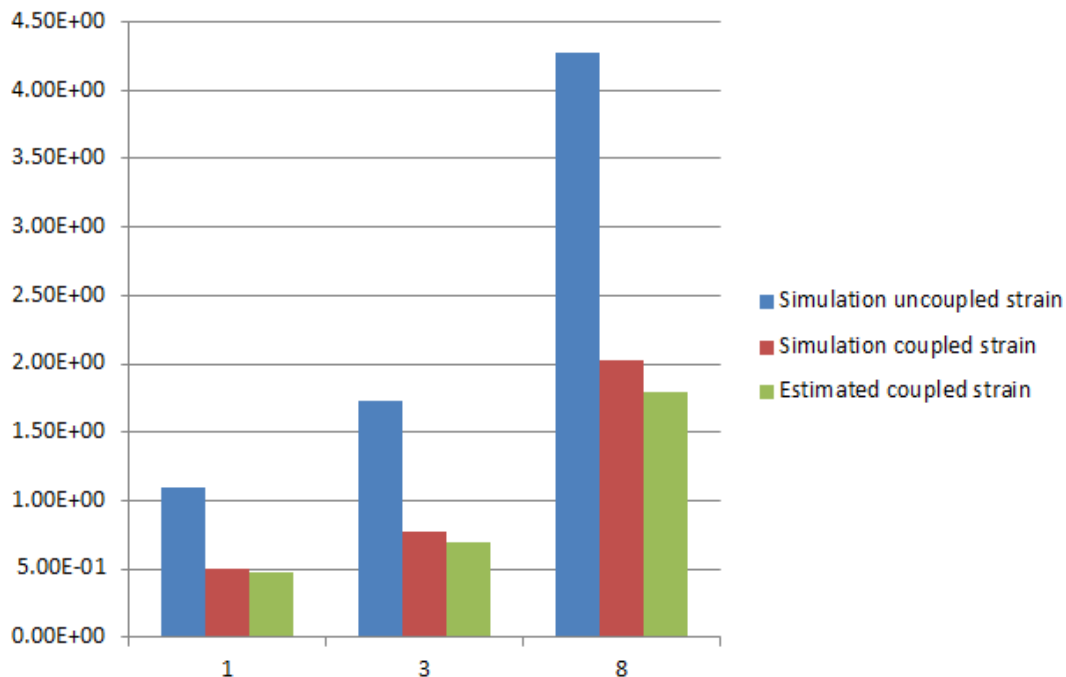
## ONE SMART MATERIAL NUMERICAL SIMULATIONS :



Neutral axis position: Error between analytical solution and simulation: **5%**

# Recent work

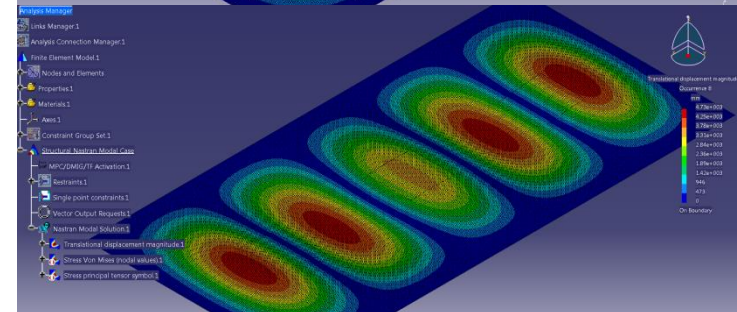
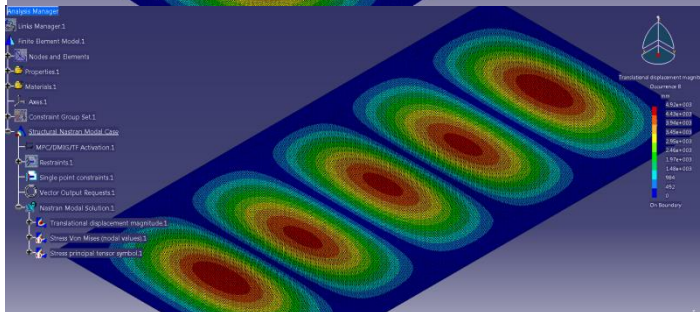
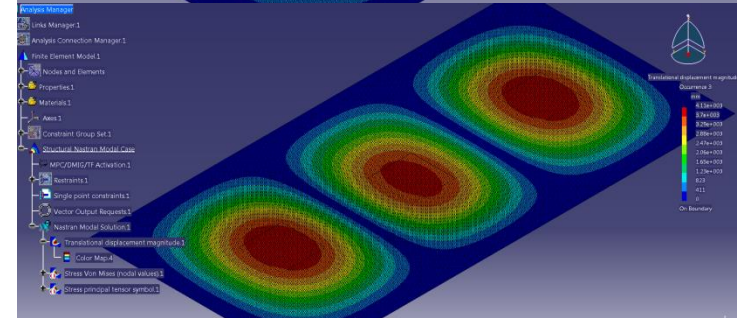
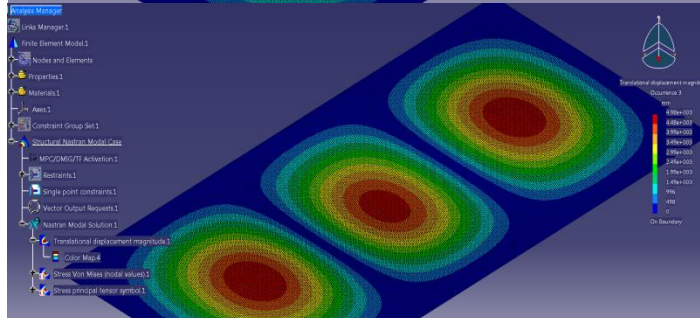
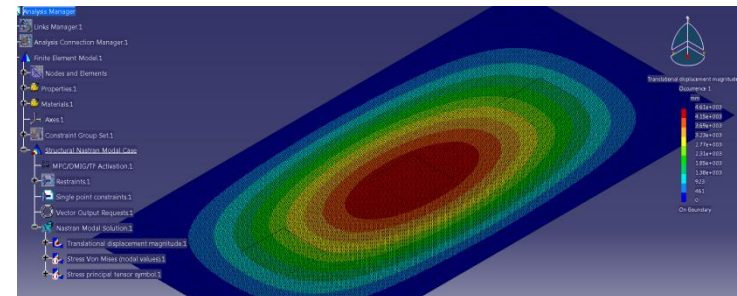
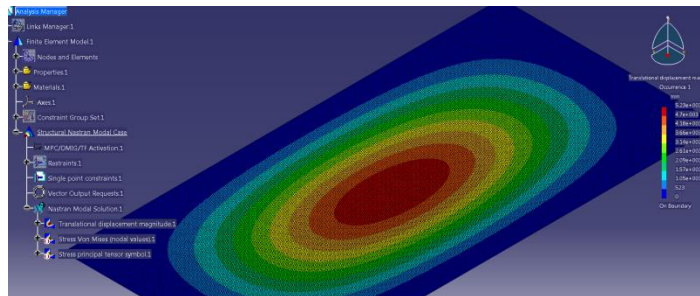
## ONE SMART MATERIAL NUMERICAL SIMULATIONS :



mode	Simulation uncoupled strain	Simulation coupled strain	Estimated coupled strain	relative error
1	1.10E+00	5.08E-01	4.72E-01	6%
3	1.73E+00	7.76E-01	6.97E-01	8%
8	4.27E+00	2.03E+00	1.80E+00	10%

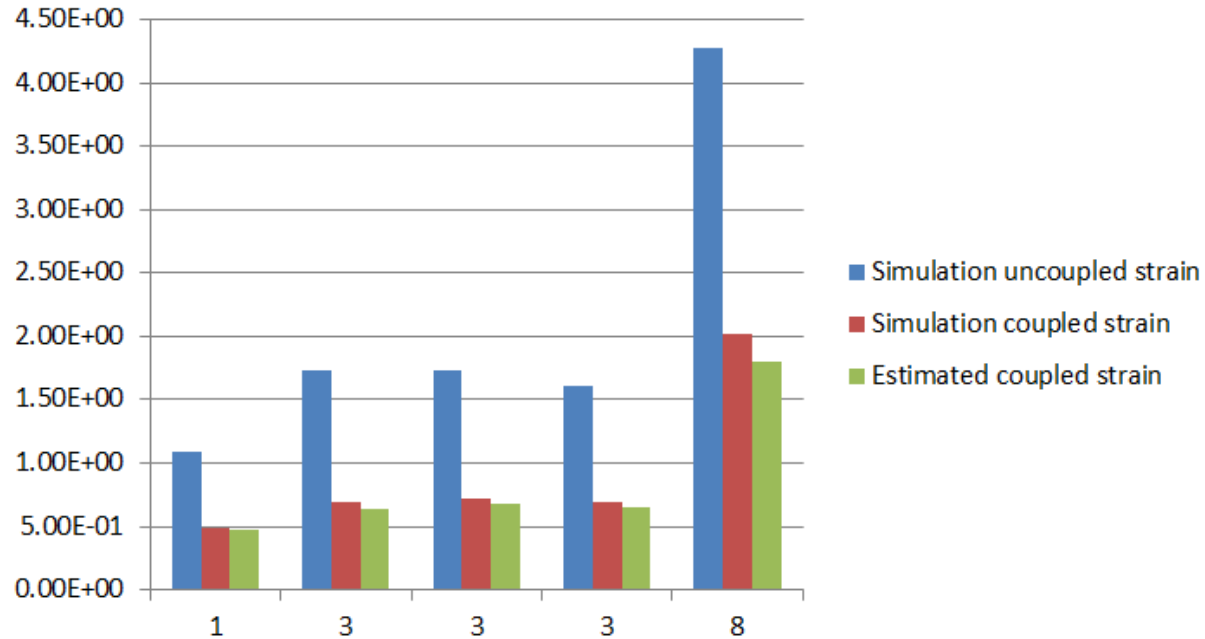
# Recent work

## THREE SMART MATERIAL NUMERICAL SIMULATIONS :



# Recent work

## THREE SMART MATERIAL NUMERICAL SIMULATIONS :



mode	Simulation uncoupled strain	Simulation coupled strain	Estimated coupled strain	relative error
1	1.10E+00	4.93E-01	4.72E-01	3%
3	1.73E+00	6.90E-01	6.35E-01	5%
3	1.73E+00	7.21E-01	6.82E-01	4%
3	1.61E+00	6.90E-01	6.56E-01	4%
8	4.27E+00	2.02E+00	1.80E+00	10%

## Future activities and difficulties

### WORK TO ACHIEVE:

1. Piezoelectric-mechanic relation of smart materials (literature search)
2. More detail FEM simulations
3. Results comparison with the literature references